

Building the Nation's Next Generation Operational Polar-Orbiting Weather Satellite

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Poster #675

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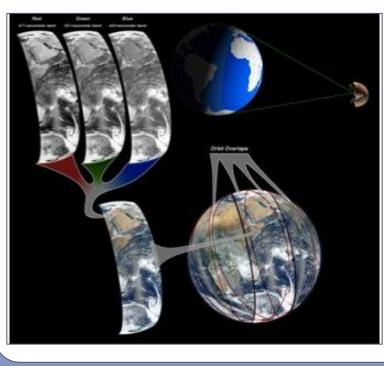
What is JPSS-1?

JPSS-1 is the Nation's next-generation, operational, polar-orbiting weather satellite being procured by NASA Goddard Space Flight Center on behalf of NOAA. The JPSS-1 spacecraft bus and its five instrument payloads are currently in full-scale production. Ball Aerospace is the provider of the JPSS-1 spacecraft bus, the Ozone Mapping and Profiler Suite (OMPS) payload and is the satellite integrator. JPSS-1 follows the successful Suomi National Polar-orbiting Partnership (S-NPP) satellite launched in October 2011. JPSS-1 will replace S-NPP after launch in the 2nd quarter of GFY 2017.

Keeping Americans safe from extreme weather events through storm tracking, enhanced weather prediction capabilities, and long-term climate monitoring is the cornerstone of the JPSS mission. The environmental monitoring which JPSS-1 will provide will advance weather forecasting and environmental prediction in many sectors, improving the ability of the public, Government, first responders, and businesses to plan for the future.

What is a Polar Orbit? Why Use a Polar Orbit?

- Polar orbits are ~90 degree inclination orbits, useful for spacecraft that carry out mapping or surveillance operations.
- ❖ Since the orbital plane is nominally fixed in inertial space, the planet rotates below a polar orbit, allowing the spacecraft low-altitude access to virtually every point on the surface.
- ❖ A polar orbit offers daily global coverage, by making ~14 polar orbits daily.





JPSS-1 Spacecraft Zenith Deck Layout (Launch configuration) **Ka-band SMD TDRSS Link Star Trackers Processors** Command & Data **Command Telemetry Power Control & Solid State Recorder Distribution Units Ka-band SMD Ground Link Batteries** TDRSS & GPS **Propulsion Module & Launch Vehicle Interface**



JPSS-1 is an *Enhancement* of the S-NPP Design

- ❖ Ball BCP-2000 spacecraft bus
- Planned launch 2Q GFY 2017
- 7-year mission life requirement
- ❖ Orbit: 824 km, sun-synch (98.7 deg), 1330L ascending node
- Observatory Mass (MEV): 1979 kg (dry)
- ❖ S/C Power (MEV): 1619 W (orbit average)
- ❖ 1553 & SpaceWire data networks
- Hydrazine propulsion system
- ❖ ADCS: 3-axis stabilized
- S-band Command & Telemetry
- * Ka-band 300 Mbps Stored Mission Data link to ground network
- ❖ Ka-band 300 Mbps Stored Mission Data back-up link to TDRSS
- ❖ X-band High Rate Data direct broadcast @ 15 Mbps
- Same payload instrument complement as S-NPP

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<u>Advanced Payloads Provide Critical Operational</u> Weather Data & Long-Term Climate Observations

Visible/Infrared Imager Radiometer Suite (VIIRS)

 A scanning radiometer, collects visible and infrared imagery and radiometric measurements of the land, atmosphere, cryosphere, and oceans

Cross-track Infrared Sounder (CrIS)

 A Fourier transform spectrometer with 1305 spectral channels will produce high-resolution, three-dimensional temperature, pressure, and moisture profiles

Advanced Technology Microwave Sounder (ATMS)

 A cross-track scanner with 22 channels, provides sounding observations needed to retrieve profiles of atmospheric temperature and moisture

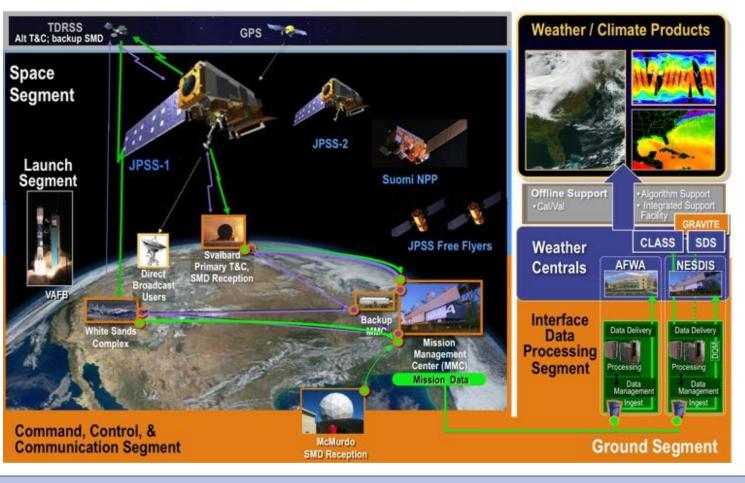
Ozone Mapping and Profiler Suite (OMPS)

 An advanced suite of hyperspectral instruments, extends the 25-plus year total-ozone and ozone-profile records

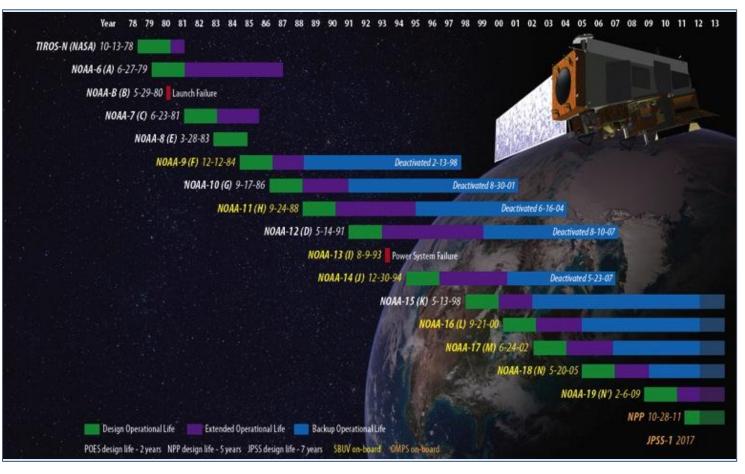
Clouds and the Earth's Radiant Energy System (CERES)

 A three-channel radiometer, measures both solarreflected and Earth-emitted radiation from the top of the atmosphere to the surface

Joint Polar Satellite System Mission Architecture



Timeline of NOAA Polar Orbiting Weather Satellites



JPSS-1 Spacecraft Nadir Deck Layout

VIIRS

TT&C Antenna

CrIS

HRD Direct Broadcast Antenna

OMPS